

# Position, Velocity, Attitude and Gyro-Bias Estimation from IMU and Position Information

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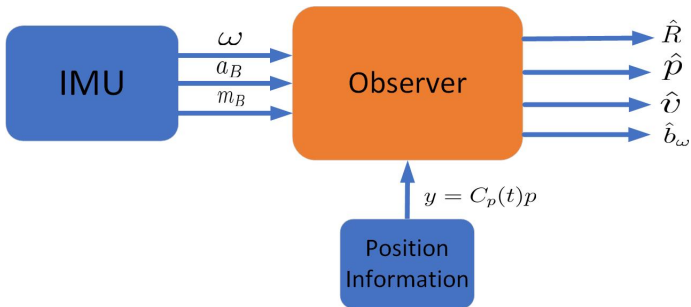
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# Problem Formulation



$$\begin{aligned}\dot{p} &= v, \\ \dot{v} &= g + Ra_B, \\ \dot{R} &= R[\omega]_\times,\end{aligned}$$



- ▶ Apparent acceleration  $a_I = Ra_B$  is unknown  $\rightarrow$  we need to estimate it as well for use in the attitude observer

# Translational Dynamics

Let  $x := (p, v, a_I) \in \mathbb{R}^9$ .

$$\dot{x} = Ax + B_1 g + B_2 \dot{a}_I$$

$$y = C(t)x$$

$$A = \begin{bmatrix} 0 & I & 0 \\ 0 & 0 & I \\ 0 & 0 & 0 \end{bmatrix}, B_1 := \begin{bmatrix} 0 \\ I \\ 0 \end{bmatrix}, B_2 := \begin{bmatrix} 0 \\ 0 \\ I \end{bmatrix}, C(t) = \begin{bmatrix} C_p(t) \\ 0 \\ 0 \end{bmatrix}^\top$$

- ▶ Linear system with unknown input  $\dot{a}_I$
- ▶ We only measure  $a_B = R^\top a_I$

## Translational Motion Observer

Consider the "quasi-linear" observer

$$\begin{cases} \dot{\hat{x}} &= \hat{z} + B_2 \hat{R} a_B \\ \dot{\hat{z}} &= A \hat{x} + B_1 g + K(t)(y - C(t)\hat{x}) + \sigma_1(\hat{R}) \end{cases}$$

with  $\sigma_1(\hat{R}) \rightarrow 0$  when  $\hat{R} \rightarrow R$  (note that  $\hat{R} a_B \rightarrow a_I$ ).

- ▶ If  $\hat{R} \equiv R$ , this is equivalent to the **Luenberger-type observer**

$$\dot{\hat{x}} = A \hat{x} + B_1 g + B_2 \dot{a}_I + K(t)(y - C(t)\hat{x}).$$

- ▶  $K(t) = \gamma L_\gamma P(t) C(t)^\top Q(t)$ , with  $\gamma \geq 1$ ,  
 $L_\gamma = \text{blockdiag}(I_3, \gamma I_3, \gamma^2 I_3)$  and  $P(t)$  is solution of the CRE

$$\frac{1}{\gamma} \dot{P} = AP + PA^\top - PC(t)^\top Q(t) C(t) P + V(t),$$

## Rotational Motion Observer

Consider the "nonlinear complementary filter"-type observer<sup>1</sup>

$$\begin{cases} \dot{\hat{R}} &= \hat{R}[\omega^y - \hat{b}_\omega + k_1\sigma_2]_\times \\ \dot{\hat{b}}_\omega &= \mathbf{Proj}(\hat{b}_\omega, -k_2\sigma_2) \end{cases}$$

$$\sigma_2 = \rho_1(m_B \times \hat{R}^\top m_I) + \rho_2(a_B \times \hat{R}^\top \mathbf{sat}(\hat{a}_I))$$

- ▶ **Proj**: is the parameter projection function
- ▶ **sat**: is a saturation function

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<sup>1</sup>R. Mahony, T. Hamel and J. Pflimlin, "Nonlinear Complementary Filters on the Special Orthogonal Group," in IEEE TAC, 2008.

H. F. Grip, T. I. Fossen, T. A. Johansen and A. Saberi, "Attitude Estimation Using Biased Gyro and Vector Measurements With Time-Varying Reference Vectors," in IEEE TAC, 2012.

# Main Result

## Assumption

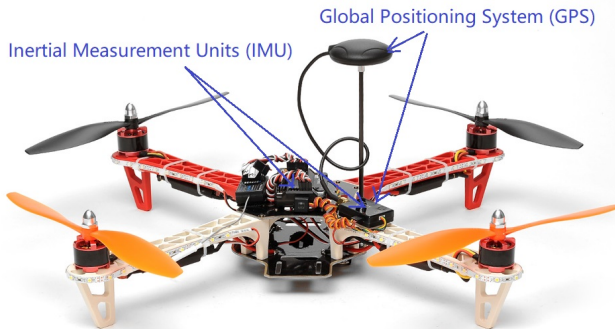
- ▶ *Pair  $(A, C(\cdot))$  is uniformly observable*
- ▶  *$m_I$  and  $a_I(t)$  are not collinear for all times*
- ▶  *$\omega, b_\omega, a_I, \dot{a}_I$  are uniformly bounded*

## Theorem (Semi-global exponential stability)

*For all initial conditions (except attitude errors at  $180^\circ$ ), there exist (high) gains such that the estimation error converges exponentially to zero.*

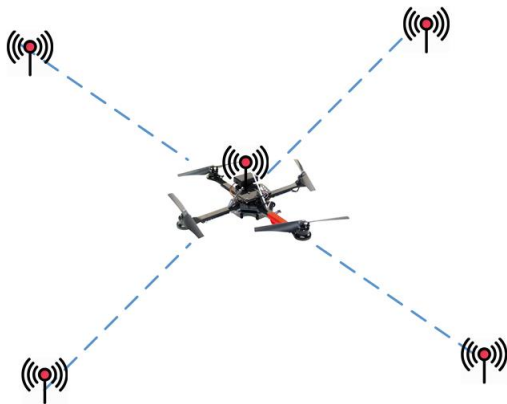
- ▶ Although, sufficiently large gains are needed in the proof, simulation results demonstrate that this is very conservative.

# Full position measurement in the inertial frame (e.g., GPS)



$$C_p = I_{3 \times 3}$$

## Range measurements (e.g., Ultra-wide-Band (UWB))





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At least 4 non-coplanar source points are needed located at  $p_i$ .

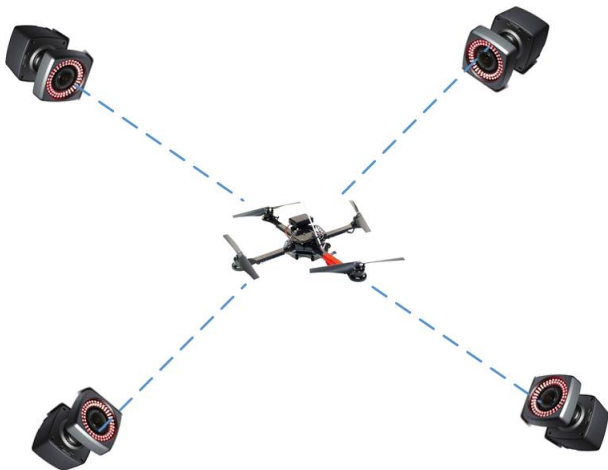
$$d_i = \|p - p_i\|, \quad i = 1, \dots, n.$$

Output equation:

$$y_i := \frac{1}{2} (d_i^2 - d_1^2 - \|p_i\|^2 + \|p_1\|^2), \quad i = 2, \dots, n.$$

$$y = \begin{bmatrix} (p_1 - p_2)^\top \\ \vdots \\ (p_1 - p_n)^\top \end{bmatrix} p := C_p p \quad (1)$$

## Bearing measurements in the inertial frame



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$$b_i = R_i^\top \frac{p - p_i}{\|p - p_i\|}, \quad i = 1, \dots, n,$$

- ▶  $R_i \in \mathbb{SO}(3)$ : Orientation of camera  $i$  w.r.t. the inertial frame.
- ▶  $p_i$ : Position of camera  $i$  in the inertial frame.
- ▶ Note that

$$\Pi(b_i)R_i^\top(p - p_i) = 0$$

where  $\Pi(z) := I - \frac{zz^\top}{\|z\|^2}$  is the orthogonal projection.

- ▶ Therefore,

$$y_i := \Pi(b_i)R_i^\top p_i = \Pi(b_i)R_i^\top p$$

and letting  $y = (y_1, \dots, y_n)$  we have

$$y = \begin{bmatrix} \Pi(b_1)R_1^\top \\ \vdots \\ \Pi(b_n)R_n^\top \end{bmatrix} p := C_p(t)p.$$

# Bearing measurements

## Lemma (Uniform observability)

$(A, C(t))$  is uniformly observable if there exist  $\delta, \mu > 0$  such that for all  $t \geq 0$  one has

$$\frac{1}{\delta} \int_t^{t+\delta} \sum_{i=1}^n \Pi(R_i b_i(s)) ds \geq \mu I_3.$$

- ▶ **Single camera:** Vehicle is never static nor moving on a straight line passing through the camera.
- ▶ **Two cameras:** Vehicle not aligned (indefinitely) with both cameras.
- ▶ **Three cameras or more:** If at least 3 cameras are not aligned, the observability condition is always satisfied.

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# Altimeter

- ▶ An altimeter is a sensor that measures the altitude

$$y = [0 \ 0 \ 1]p.$$

## Lemma (Uniform observability: bearings+altimeter)

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$$\frac{1}{\delta} \int_t^{t+\delta} \sum_{i=1}^n \Pi(R_i b_i(s)) ds + e_3 e_3^\top \geq \mu I_3.$$

- ▶ **Single camera + altimeter:** Vehicle not moving at the same altitude as the camera or, if doing so, not moving on a straight line passing through the camera
- ▶ **Two cameras + altimeter:** Cameras not at the same altitude. If the cameras are at the same altitude, vehicle must not be aligned (indefinitely) with both cameras.

## Simulations: true trajectory

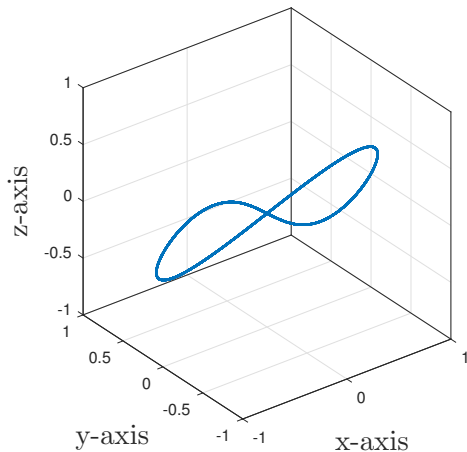


Figure 1: True trajectory of the vehicle.



## Simulations: single bearing

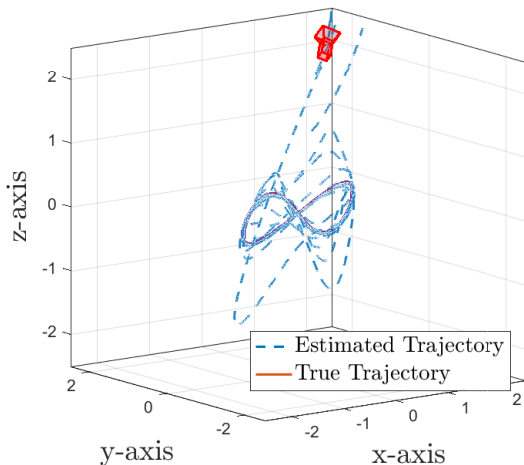


Figure 2: Observability holds but slow convergence.

## Simulations: single bearing + altimeter

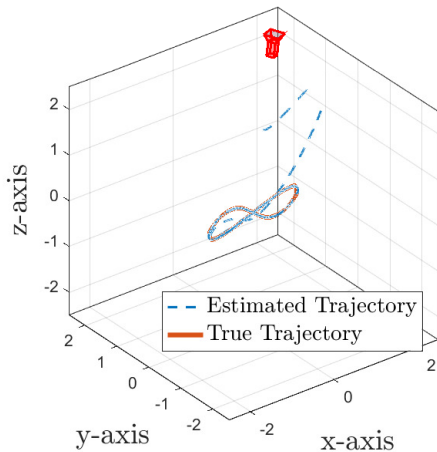


Figure 3: Stronger observability.

# Conclusion

- ▶ A **generic** navigation observer with semi-global exponential stability
- ▶ Position information (full, bearings, range) in the inertial frame
- ▶ Future work using information in the body-frame (e.g., on-board cameras)

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**Thank you**

**Questions?**